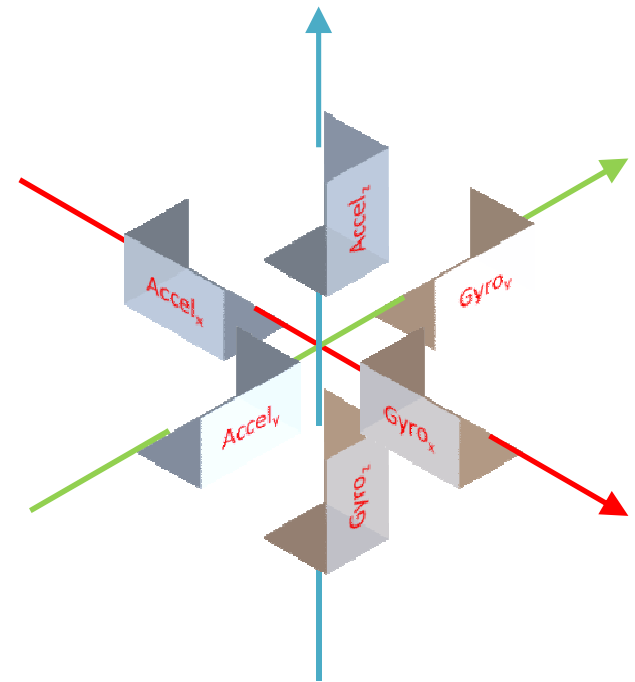
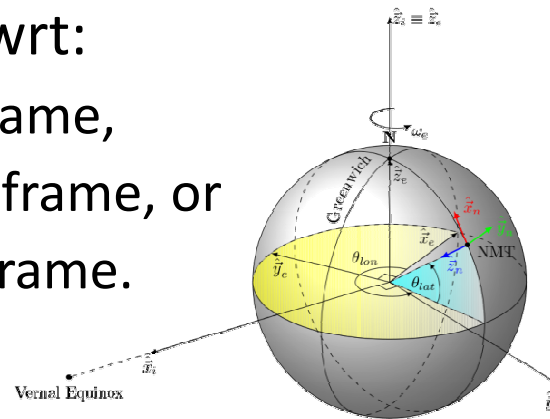


Inertial Navigation

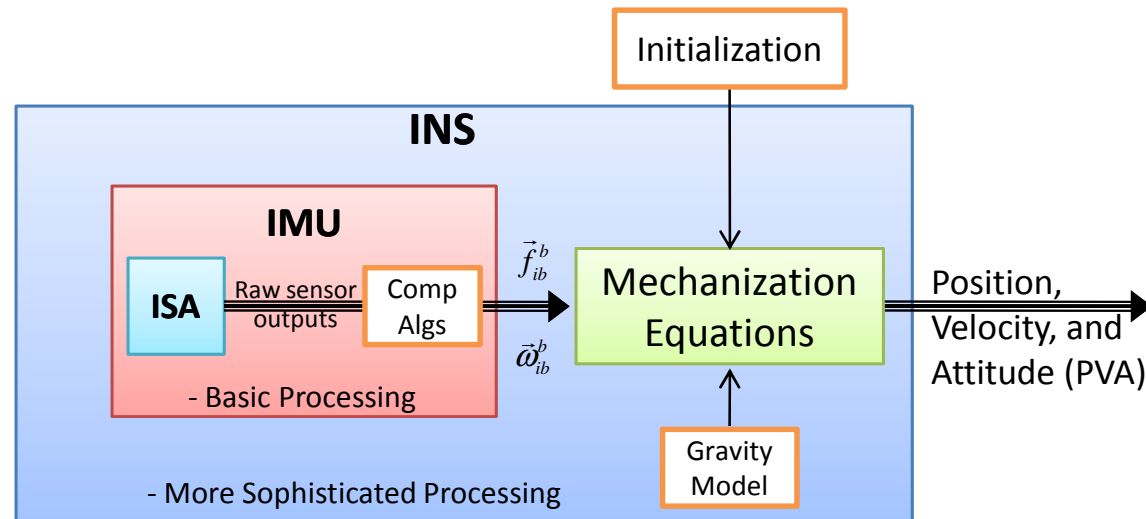
- Inertial Navigation

- The process of “integrating” angular velocity & acceleration to determine One’s position, velocity, and attitude (PVA)
 - Effectively “dead reckoning”
- To measure the acceleration and angular velocity vectors we need at least 3-gyros and 3-accelers
 - Typically configured in an orthogonal triad
- The “mechanization” can be performed wrt:
 - The ECI frame,
 - The ECEF frame, or
 - The Nav frame.



Inertial Navigation

- An Inertial Navigation System (INS)
 - ISA – Inertial Sensor Assembly
 - Typically, 3-gyros + 3-accel + basic electronics (power, ...)
 - IMU – Inertial Measurement Unit
 - ISA + Compensation algorithms (i.e. basic processing)
 - INS – Inertial Navigation System
 - IMU + gravity model + “mechanization” algorithms



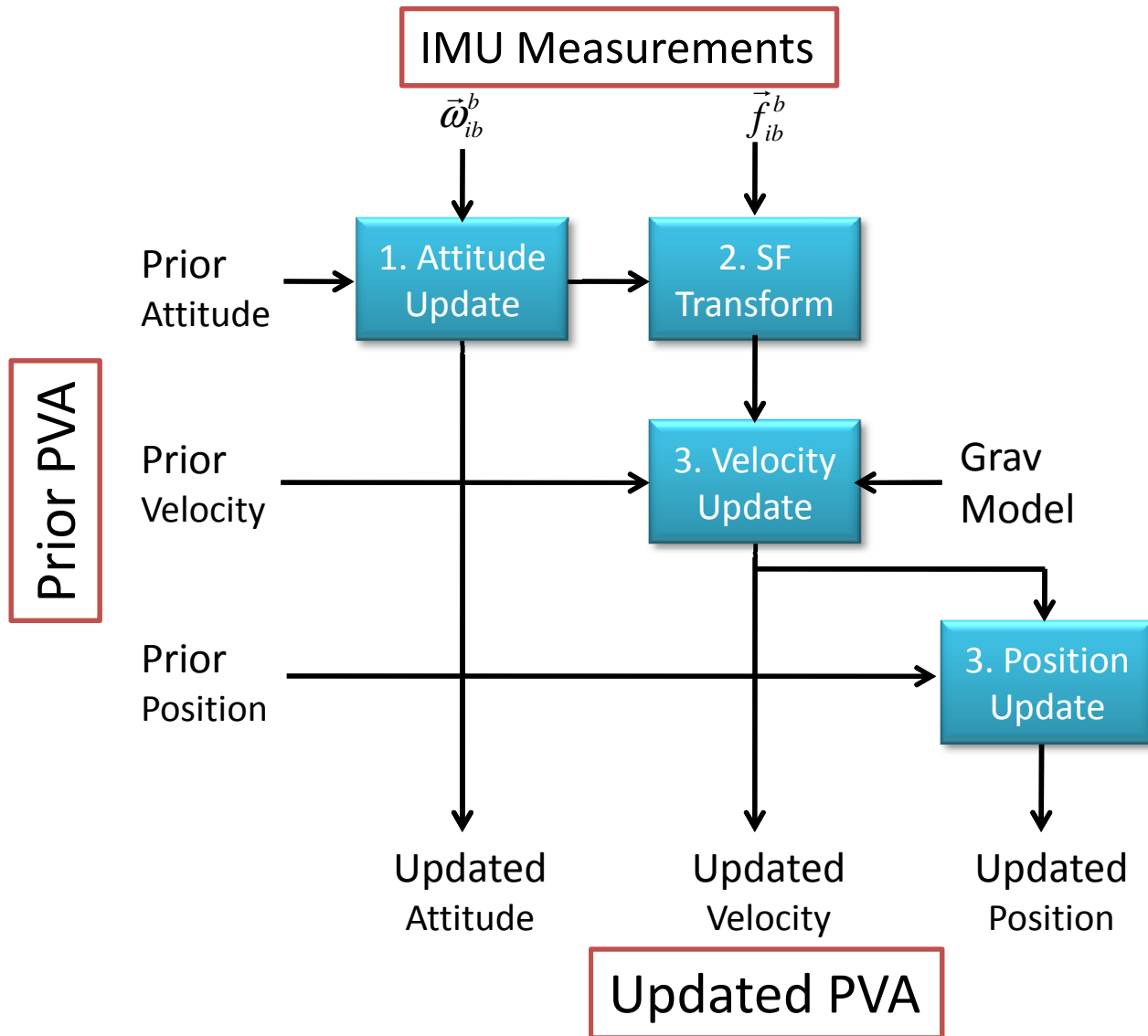
Inertial Navigation

A Four Step Mechanization

- Can be generically performed in four steps:
 1. Attitude Update
 - Update the prior attitude (a DCM, say) using the current angular velocity measurement
 2. Transform the specific force measurement
 - Typically, using the attitude computed in step 1.
 3. Update the velocity
 - Essentially integrate the result from step 2. with the use of a gravity model
 4. Update the Position
 - Essentially integrate the result from step 3.

Inertial Navigation

A Four Step Mechanization



Inertial Navigation

Case 1: ECI Frame Mechanization

- Determine our PVA wrt the ECI frame

- Namely: Position \vec{r}_{ib}^i , Velocity, \vec{v}_{ib}^i , and attitude C_b^i

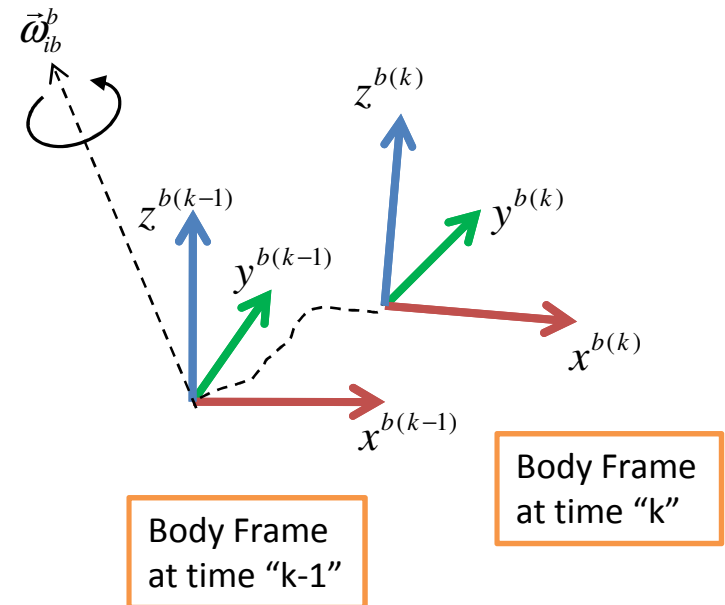
1. Attitude Update:

- Body orientation frame at time “k” wrt time “k-1”
 - $\Delta t = \text{Time } k - \text{Time } k-1$

$$C_{b(k)}^i = C_{b(k-1)}^i C_{b(k)}^{b(k-1)}$$

$$C_{b(k)}^{b(k-1)} = e^{\Omega_{ib}^b \Delta t}$$

$$\begin{aligned} C_b^i(+)\Delta t &= C_b^i(-) e^{\Omega_{ib}^b \Delta t} \\ &\simeq C_b^i(-) \left(I + [\vec{\omega}_{ib}^b \times] \Delta t \right) \end{aligned}$$



Inertial Navigation

Case 1: ECI Frame Mechanization

2. Specific Force Transformation

- Simply coordinatize the specific force

$$\vec{f}_{ib}^i = C_b^i(+)\vec{f}_{ib}^b$$

3. Velocity Update

- Assuming that we are in space (*i.e.* no centrifugal component)

$$\vec{a}_{ib}^i = \vec{f}_{ib}^i - \vec{\gamma}_{ib}^i$$

- Thus, by simple numerical integration

$$\vec{v}_{ib}^i(+)=\vec{v}_{ib}^i(-)+\vec{a}_{ib}^i\Delta t$$

4. Position Update

- By simple numerical integration

$$\vec{r}_{ib}^i(+)=\vec{r}_{ib}^i(-)+\vec{v}_{ib}^i(-)\Delta t+\vec{a}_{ib}^i\frac{\Delta t^2}{2}$$

Inertial Navigation

Case 1: ECI Frame Mechanization

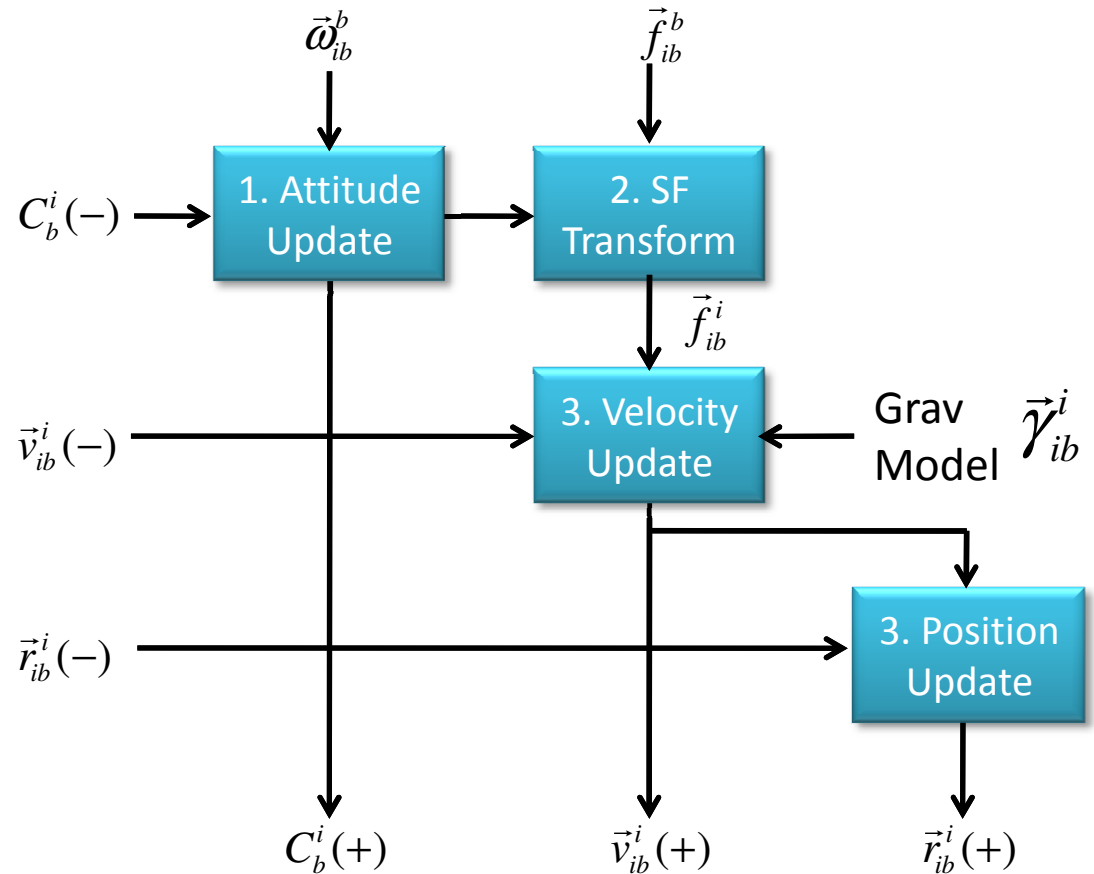
$$C_b^i(+)=C_b^i(-)\left(I+\left[\vec{\omega}_{ib}^b\times\right]\Delta t\right)$$

$$\vec{f}_{ib}^i=C_b^i(+)\vec{f}_{ib}^b$$

$$\vec{a}_{ib}^i=\vec{f}_{ib}^i-\vec{\gamma}_{ib}^i$$

$$\vec{v}_{ib}^i(+)=\vec{v}_{ib}^i(-)+\vec{a}_{ib}^i\Delta t$$

$$\vec{r}_{ib}^i(+)=\vec{r}_{ib}^i(-)+\vec{v}_{ib}^i(-)\Delta t+\vec{a}_{ib}^i\frac{\Delta t^2}{2}$$



Inertial Navigation

Case 2: ECEF Frame Mechanization

- Determine our PVA wrt the ECI frame

- Namely: Position \vec{r}_{eb}^e , Velocity, \vec{v}_{eb}^e , and attitude C_b^e

1. Attitude Update:

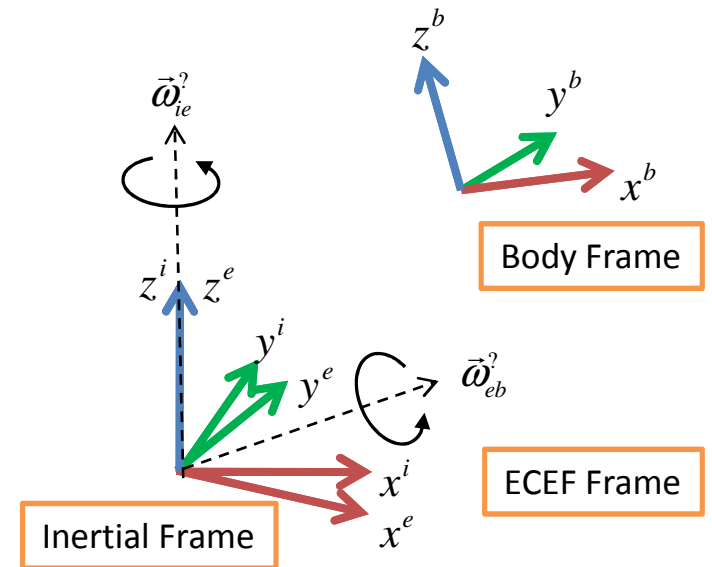
- Start with the angular velocity $\vec{\omega}_{ib}^? = \vec{\omega}_{ie}^? + \vec{\omega}_{eb}^?$

$$\vec{\omega}_{ib}^e = \vec{\omega}_{ie}^e + \vec{\omega}_{eb}^e$$

$$\vec{\omega}_{eb}^e = C_b^e \vec{\omega}_{ib}^b - \vec{\omega}_{ie}^i \quad [(C\omega)\times] = C[\omega\times]C^T$$

$$\Omega_{eb}^e = C_b^e \Omega_{ib}^b C_e^b - \Omega_{ie}^i$$

$$\begin{aligned} C_b^e(+)\Delta t &= e^{\Omega_{eb}^e \Delta t} C_b^e(-) \\ &\approx \left(I + C_b^e \Omega_{ib}^b C_e^b \Delta t - \Omega_{ie}^i \Delta t \right) C_b^e(-) \\ &= C_b^e(-) \left[I + \Omega_{ib}^b \Delta t \right] - \Omega_{ie}^i C_b^e(-) \Delta t \end{aligned}$$



Inertial Navigation

Case 2: ECEF Frame Mechanization

2. Specific Force Transformation:

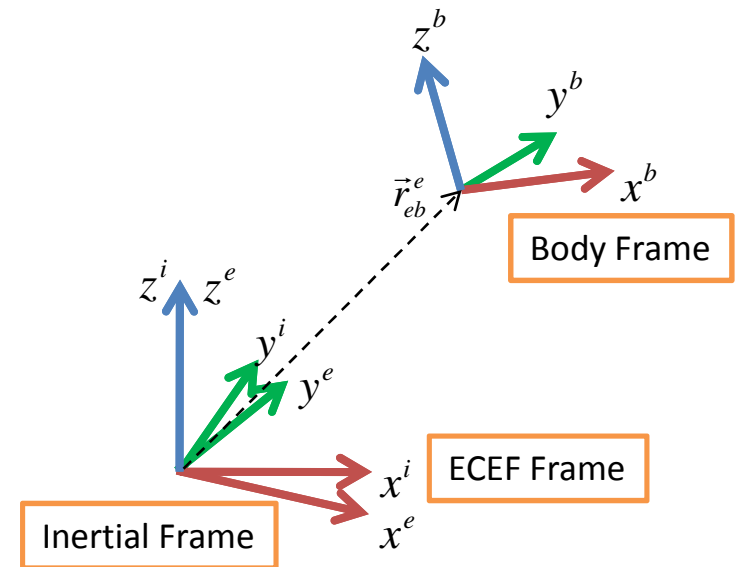
- Simply coordinatize the specific force

$$\vec{f}_{ib}^e = C_b^e (+) \vec{f}_{ib}^b$$

3. Velocity Update

- ECEF & ECI have the same origin

$$\begin{aligned} \vec{r}_{ib}^i &= \vec{r}_{ie}^i + C_e^i \vec{r}_{eb}^e \Rightarrow \vec{r}_{eb}^e = C_i^e \vec{r}_{ib}^i \\ \vec{v}_{eb}^e &= \dot{\vec{r}}_{eb}^e \\ &= \dot{C}_i^e \vec{r}_{ib}^i + C_i^e \dot{\vec{r}}_{ib}^i \\ &= \Omega_{ei}^e C_i^e \vec{r}_{ib}^i + C_i^e \vec{v}_{ib}^i = -\Omega_{ie}^e \vec{r}_{eb}^e + C_i^e \vec{v}_{ib}^i \end{aligned}$$



Inertial Navigation

Case 2: ECEF Frame Mechanization

$$\vec{a}_{eb}^e = \dot{\vec{v}}_{eb}^e = \frac{d}{dt} \left(-\Omega_{ie}^e \vec{r}_{eb}^e + C_i^e \vec{v}_{ib}^i \right)$$

$$= -\Omega_{ie}^e \dot{\vec{r}}_{eb}^e + \dot{C}_i^e \vec{r}_{ib}^i + C_i^e \dot{\vec{v}}_{ib}^i$$

$$= -\Omega_{ie}^e \vec{v}_{eb}^e + \Omega_{ei}^e \boxed{C_i^e \vec{v}_{ib}^i} + C_i^e \vec{a}_{ib}^i$$

$$= -\Omega_{ie}^e \vec{v}_{eb}^e - \Omega_{ie}^e \left[\vec{v}_{eb}^e + \Omega_{ie}^e \vec{r}_{eb}^e \right] + C_i^e \vec{a}_{ib}^i$$

$$= -2\Omega_{ie}^e \vec{v}_{eb}^e - \Omega_{ie}^e \Omega_{ie}^e \vec{r}_{eb}^e + \boxed{\vec{a}_{ib}^e}$$

$$= -2\Omega_{ie}^i \vec{v}_{eb}^e + \vec{f}_{ib}^e + \vec{g}_b^e$$

$$C_i^e \vec{v}_{ib}^i = \Omega_{ie}^e \vec{r}_{eb}^e + \vec{v}_{eb}^e$$

$$\vec{f}_{ib}^? = \vec{a}_{ib}^? - \gamma_{ib}^?$$

$$\vec{a}_{ib}^e = \vec{f}_{ib}^e + \boxed{\gamma_{ib}^e}$$

$$\vec{g}_b^e = \gamma_{ib}^e - \Omega_{ie}^e \Omega_{ie}^e \vec{r}_{eb}^e$$

$$\vec{a}_{ib}^e = \vec{f}_{ib}^e + \vec{g}_b^e + \Omega_{ie}^e \Omega_{ie}^e \vec{r}_{eb}^e$$

$$\vec{v}_{eb}^e (+) = \vec{v}_{eb}^e (-) + \vec{a}_{eb}^e \Delta t$$

$$= \vec{v}_{eb}^e (-) + \left[\vec{f}_{ib}^e + \vec{g}_b^e - 2\Omega_{ie}^i \vec{v}_{eb}^e (-) \right] \Delta t$$

Inertial Navigation

Case 2: ECEF Frame Mechanization

4. Position Update

- By simple numerical integration

$$\vec{r}_{eb}^e(+)=\vec{r}_{eb}^e(-)+\vec{v}_{eb}^e(-)\Delta t+\vec{a}_{eb}^e\frac{\Delta t^2}{2}$$

Inertial Navigation

Case 2: ECEF Frame Mechanization

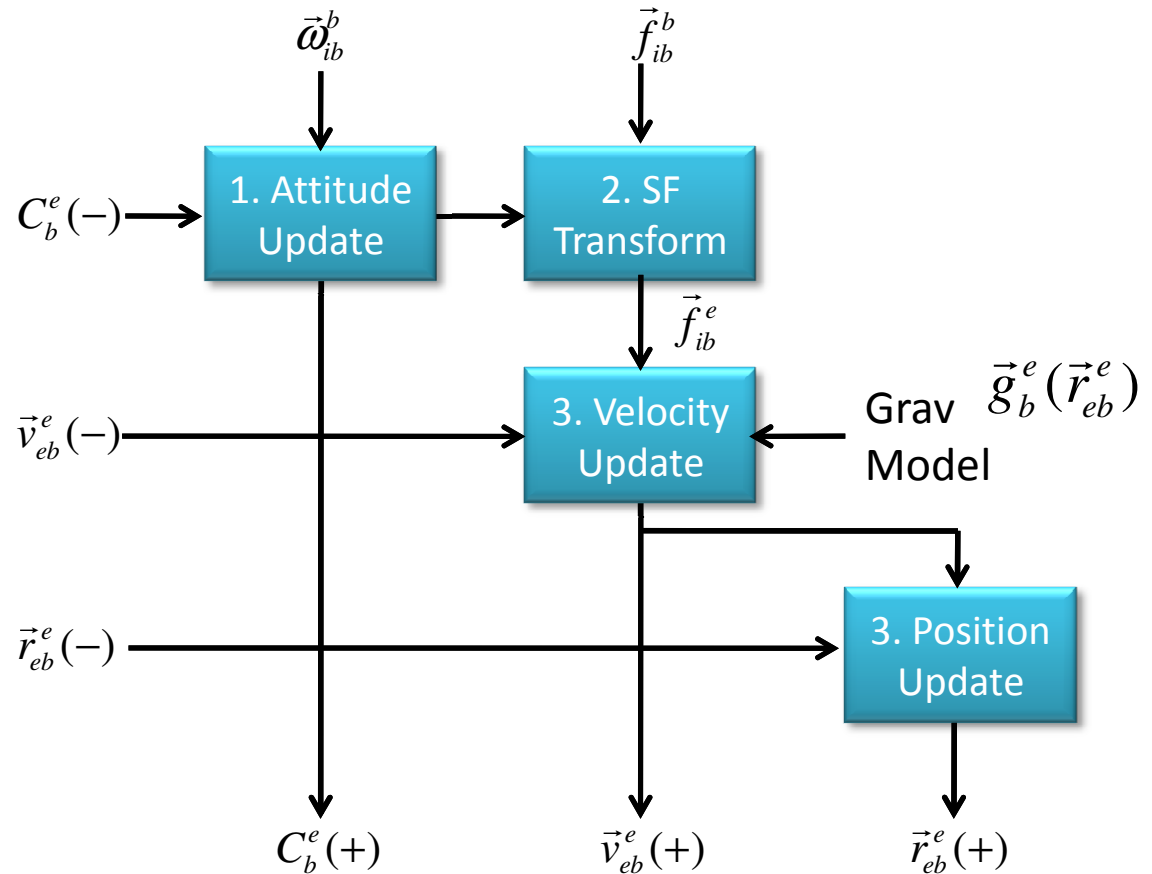
$$C_b^e(+)=C_b^e(-)\left[I+\Omega_{ib}^b\Delta t\right]-\Omega_{ie}^iC_b^e(-)\Delta t$$

$$\vec{f}_{ib}^e=C_b^e(+)\vec{f}_{ib}^b$$

$$\vec{a}_{eb}^e=\vec{f}_{ib}^e+\vec{g}_b^e-2\Omega_{ie}^i\vec{v}_{eb}^e(-)$$

$$\vec{v}_{eb}^e(+)=\vec{v}_{eb}^e(-)+\vec{a}_{eb}^e\Delta t$$

$$\vec{r}_{eb}^e(+)=\vec{r}_{eb}^e(-)+\vec{v}_{eb}^e(-)\Delta t+\vec{a}_{eb}^e\frac{\Delta t^2}{2}$$



Inertial Navigation

Case 3: Navigation Frame Mechanization

- Determine our PVA wrt the Nav frame
 - Position: Typically described in curvilinear coordinates

$$[L_b, \lambda_b, h_b]^T$$

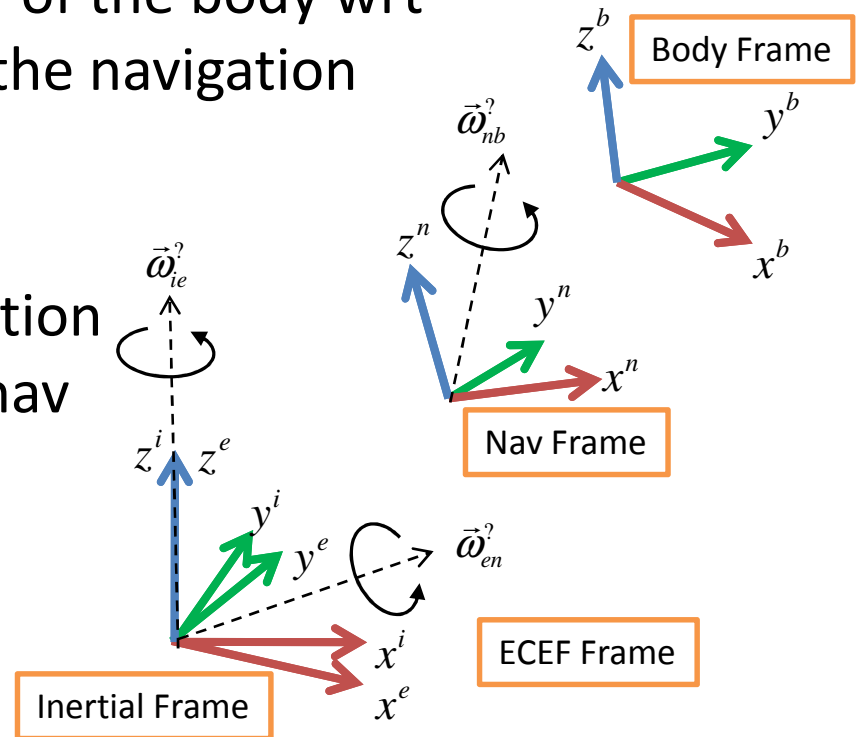
- Velocity: Typically the velocity of the body wrt the earth frame described in the navigation frame coords

$$\vec{v}_{eb}^n$$

- Attitude: Typically the orientation of the body described in the nav frame

$$C_b^n$$

- This unusual choice facilitates the guidance system function



Inertial Navigation

Case 3: Navigation Frame Mechanization

1. Attitude Update:

- Note that $\vec{\omega}_{ib}^? = \vec{\omega}_{ie}^? + \vec{\omega}_{en}^? + \vec{\omega}_{nb}^? \Rightarrow \vec{\omega}_{nb}^b = \vec{\omega}_{ib}^b - \vec{\omega}_{ie}^b - \vec{\omega}_{en}^b$

- Now $\dot{C}_b^n = C_b^n \Omega_{nb}^b = C_b^n (\Omega_{ib}^b - \Omega_{ie}^b - \Omega_{en}^b)$

$$= C_b^n \Omega_{ib}^b - C_b^n \Omega_{ie}^b - C_b^n \Omega_{en}^b$$

$$= C_b^n \Omega_{ib}^b - (\Omega_{ie}^n + \Omega_{en}^n) C_b^n$$

See next slide

Measured by the gyro

$$\vec{\omega}_{ie}^n = C_e^n \vec{\omega}_{ie}^e$$

$$= \begin{bmatrix} * & * & * \\ * & * & * \\ \cos(L_b) & 0 & -\sin(L_b) \end{bmatrix}^T \begin{bmatrix} 0 \\ 0 \\ \omega_{ie} \end{bmatrix} = \omega_{ie} \begin{bmatrix} \cos(L_b) \\ 0 \\ -\sin(L_b) \end{bmatrix}$$

Inertial Navigation

Case 3: Navigation Frame Mechanization

- Last term: $\Omega_{en}^n \Leftrightarrow \vec{\omega}_{en}^n$

Courtesy of Mathematica

$$\dot{C}_n^e = C_n^e \Omega_{en}^n \Rightarrow \Omega_{en}^n = (C_n^e)^T \dot{C}_n^e = \begin{pmatrix} 0 & \sin(L_b) \dot{\lambda}_b & \omega_{en,y}^n \\ -\sin(L_b) \dot{\lambda}_b & 0 & -\cos(L_b) \dot{\lambda}_b \\ \dot{L}_b & \cos(L_b) \dot{\lambda}_b & 0 \end{pmatrix}$$

$$\vec{\omega}_{en}^n = \begin{bmatrix} \cos(L_b) \dot{\lambda}_b \\ -\dot{L}_b \\ -\sin(L_b) \dot{\lambda}_b \end{bmatrix}$$

From Eqn. 16 (Wedeward handout) \Rightarrow

$$\begin{pmatrix} \dot{L}_b \\ \dot{\lambda}_b \\ \dot{h}_b \end{pmatrix} = \begin{pmatrix} \frac{\vec{v}_{eb,N}^n}{(R_N + h_b)} \\ \frac{\vec{v}_{eb,E}^n}{\cos(L_b)(R_E + h_b)} \\ -\vec{v}_{eb,D}^n \end{pmatrix}$$

$$\therefore \vec{\omega}_{en}^n = \begin{bmatrix} \frac{\vec{v}_{eb,E}^n}{(R_E + h_b)} \\ -\frac{\vec{v}_{eb,N}^n}{(R_N + h_b)} \\ -\frac{\tan(L_b) \vec{v}_{eb,E}^n}{(R_E + h_b)} \end{bmatrix}$$

$$\begin{aligned} C_b^n (+) &\approx C_b^n (-) + \Delta t \dot{C}_b^n \\ &= C_b^n (-) (I + \Delta t \Omega_{ib}^b) - (\Omega_{ie}^n + \Omega_{en}^n) C_b^n (-) \Delta t \end{aligned}$$

Inertial Navigation

Case 3: Nav Frame Mechanization

2. Specific Force Transformation:

- Simply coordinatize the specific force

$$\vec{f}_{ib}^n = C_b^n (+) \vec{f}_{ib}^b$$

3. Velocity Update

- Recall that $[(C\omega)\times] = C[\omega\times]C^T \Rightarrow C[\omega\times] = [(C\omega)\times]C$

$$\vec{v}_{eb}^n = C_e^n \vec{v}_{eb}^e$$

$$\begin{aligned}\dot{\vec{v}}_{eb}^n &= \dot{C}_e^n \vec{v}_{eb}^e + C_e^n \dot{\vec{v}}_{eb}^e \\ &= \Omega_{ne}^n C_e^n \vec{v}_{eb}^e + C_e^n \left(\vec{f}_{ib}^e + \vec{g}_b^e - 2\Omega_{ie}^e \vec{v}_{eb}^e \right) \\ &= \vec{f}_{ib}^n + \vec{g}_b^n - \Omega_{en}^n \vec{v}_{eb}^n - 2C_e^n \Omega_{ie}^e \vec{v}_{eb}^e \\ &= \vec{f}_{ib}^n + \vec{g}_b^n - \Omega_{en}^n \vec{v}_{eb}^n - 2\Omega_{ie}^n C_e^n \vec{v}_{eb}^e \\ &= \vec{f}_{ib}^n + \vec{g}_b^n - \left(\Omega_{en}^n + 2\Omega_{ie}^n \right) \vec{v}_{eb}^n\end{aligned}$$

Inertial Navigation

Case 3: Nav Frame Mechanization

- Finally

$$\vec{v}_{eb}^n (+) = \vec{v}_{eb}^n (-) + \Delta t \left[\vec{f}_{ib}^n + \vec{g}_b^n - \left(\Omega_{en}^n + 2\Omega_{ie}^n \right) \vec{v}_{eb}^n (-) \right]$$

4. Position Update

- Recalling the relationship between \vec{v}_{eb}^n and the curvilinear coordinates

$$\begin{pmatrix} \dot{L}_b \\ \dot{\lambda}_b \\ \dot{h}_b \end{pmatrix} = \begin{pmatrix} \frac{\vec{v}_{eb,N}^n}{(R_N + h_b)} \\ \frac{\vec{v}_{eb,E}^n}{\cos(L_b)(R_E + h_b)} \\ -\vec{v}_{eb,D}^n \end{pmatrix}$$

$$h_b (+) = h_b (-) + \Delta t \left[\vec{v}_{eb,D}^n (+) \right]$$

$$L_b (+) = L_b (-) + \Delta t \left[\frac{\vec{v}_{eb,N}^n (+)}{R_N - h_b} \right]$$

$$\lambda_b (+) = \lambda_b (-) + \Delta t \left[\frac{\vec{v}_{eb,E}^n (+)}{(R_E - h_b) \cos(L_b)} \right]$$

Inertial Navigation

Case 3: Nav Frame Mechanization

$$C_b^n (+) = C_b^n (-) (I + \Delta t \Omega_{ib}^b) - (\Omega_{ie}^n + \Omega_{en}^n) C_b^n (-) \Delta t$$

$$\vec{f}_{ib}^n = C_b^n (+) \vec{f}_{ib}^b$$

$$\vec{v}_{eb}^n (+) = \vec{v}_{eb}^n (-) + \Delta t \left[\vec{f}_{ib}^n + \vec{g}_b^n - (\Omega_{en}^n + 2\Omega_{ie}^n) \vec{v}_{eb}^n (-) \right]$$

$$h_b (+) = h_b (-) + \Delta t \left[\vec{v}_{eb,D}^n (+) \right]$$

$$L_b (+) = L_b (-) + \Delta t \left[\frac{\vec{v}_{eb,N}^n (+)}{R_N - h_b} \right]$$

$$\lambda_b (+) = \lambda_b (-) + \Delta t \left[\frac{\vec{v}_{eb,E}^n (+)}{(R_E - h_b) \cos(L_b)} \right]$$

