

# EE 570: Location and Navigation

## Navigation Equations: ECI Mechanization

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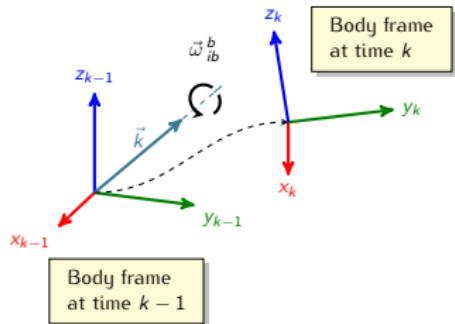
February 18, 2016

- Determine the position, velocity and attitude of the **body** frame wrt the **inertial** frame
  - **Position** — Vector from the origin of the inertial frame to the origin of the body frame resolved in the inertial frame:  $\vec{r}_{ib}^i$
  - **Velocity** — Velocity of the body frame wrt the inertial frame resolved in the inertial frame:  $\vec{v}_{ib}^i$
  - **Attitude** — Orientation of the body frame wrt the inertial frame:  $C_b^i$  or  $\vec{q}_b^i$

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  - **Attitude** — Orientation of the body frame wrt the inertial frame:  $C_b^i$  or  $\vec{q}_b^i$
- The inputs are  $\vec{\omega}_{ib}^b$  and  $\vec{f}_{ib}^b$

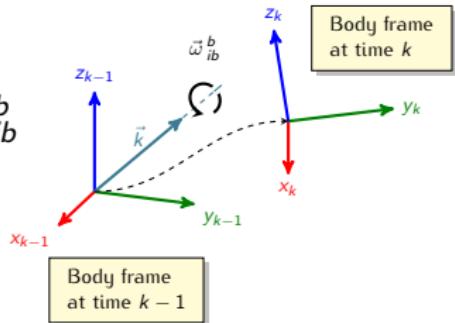
- Body orientation frame at time “ $k$ ” wrt time “ $k - 1$ ”
- $\Delta t = t_k - t_{k-1}$

$$\dot{C}_b^i = C_b^i \Omega_{ib}^b$$



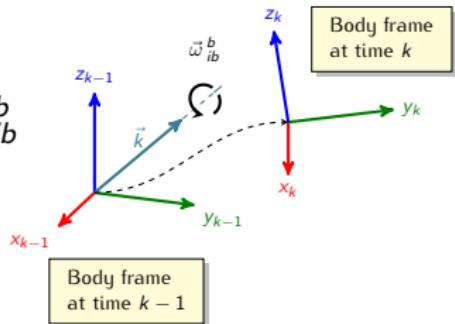
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$$\begin{aligned}\dot{C}_b^i &= C_b^i \Omega_{ib}^b \\ &= \lim_{\Delta t \rightarrow 0} \left( \frac{C_b^i(k) - C_b^i(k-1)}{\Delta t} \right) = C_b^i(k-1) \Omega_{ib}^b\end{aligned}$$



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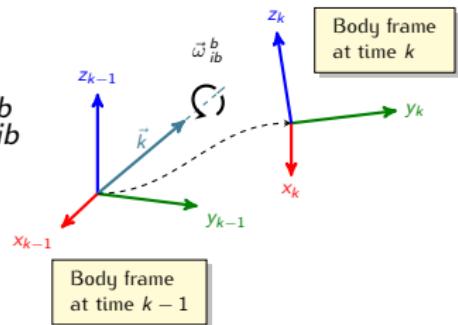


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$$C_b^i(+) - C_b^i(-) \approx C_b^i(-) \Omega_{ib}^b \Delta t$$

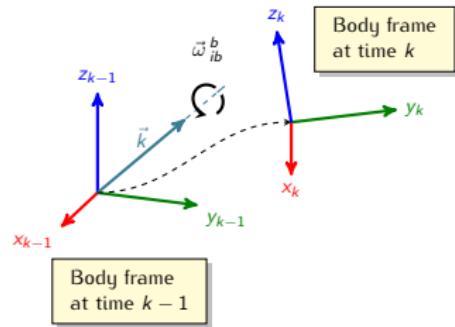
$$C_b^i(+) \approx C_b^i(-) \left( \mathcal{I} + \Omega_{ib}^b \Delta t \right)$$



- Body orientation frame at time “ $k$ ” wrt time “ $k - 1$ ”
  - $\Delta t = t_k - t_{k-1}$

$$C_{b(k)}^i = C_{b(k-1)}^i C_{b(k)}^{b(k-1)}$$

$$\vec{\omega}_{ib}^b \Delta t = \vec{k} \Delta \theta$$



Body frame  
at time  $k - 1$

Body frame  
at time  $k$

$$\mathfrak{K} = [\vec{k} \times]$$

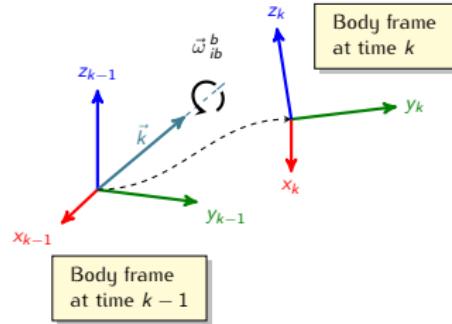
## Attitude — Method B

- Body orientation frame at time "k" wrt time "k - 1"
  - $\Delta t = t_k - t_{k-1}$

$$C_{b(k)}^i = C_{b(k-1)}^i C_{b(k)}^{b(k-1)}$$

$$\vec{\omega}_{ib}^b \Delta t = \vec{k} \Delta \theta$$

$$C_{b(k)}^{b(k-1)} = e^{\Omega_{ib}^b \Delta t} = e^{\vec{\kappa} \Delta \theta}$$



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## Attitude — Method B

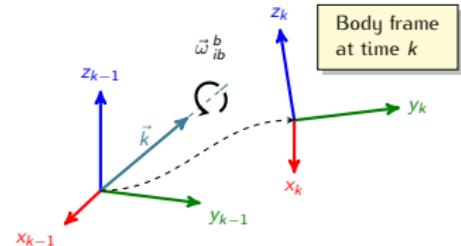
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$$= \mathcal{I} + \tilde{\kappa} \Delta \theta + \frac{\tilde{\kappa}^2 \Delta \theta^2}{2!} + \frac{\tilde{\kappa}^3 \Delta \theta^3}{3!} + \dots$$



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at time  $k-1$

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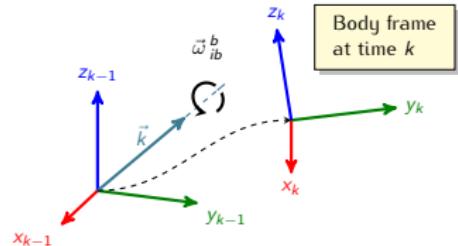
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$$= \mathcal{I} + \sin(\Delta \theta) \tilde{\kappa} + [1 - \cos(\Delta \theta)] \tilde{\kappa}^2$$



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  - $\Delta t = t_k - t_{k-1}$

$$C_{b(k)}^i = C_{b(k-1)}^i C_{b(k)}^{b(k-1)}$$

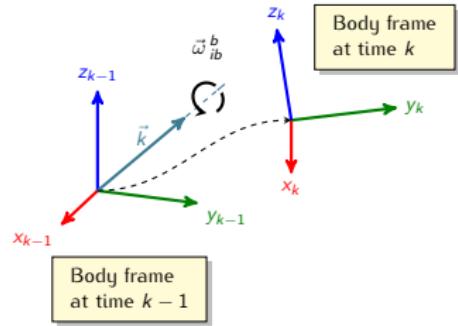
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$$C_b^i(+) = C_b^i(-) C_{b(k)}^{b(k-1)}$$



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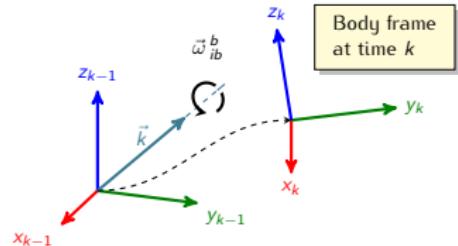
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$$\approx C_b^i(-) (\mathcal{I} + \Omega_{ib}^b \Delta t)$$



$$\text{Body frame at time } k-1$$

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- Body orientation frame at time “ $k$ ” wrt time “ $k - 1$ ”

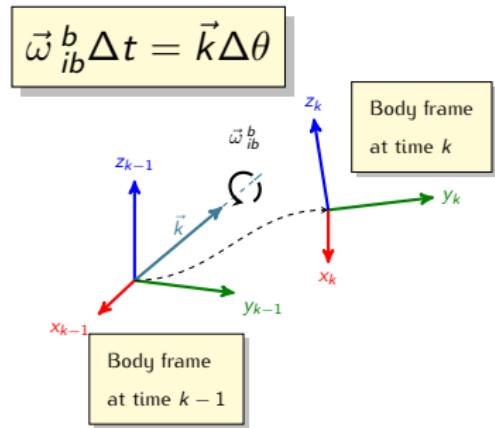
- $\Delta t = t_k - t_{k-1}$

$$\bar{q}^i_{b(k)} = \bar{q}^i_{b(k-1)} \otimes \bar{q}^{b(k-1)}_{b(k)}$$

$$\bar{q}^{b(k-1)}_{b(k)} = \begin{bmatrix} \cos(\frac{\Delta\theta}{2}) \\ \vec{k} \sin(\frac{\Delta\theta}{2}) \end{bmatrix}$$

$$\bar{q}^i_b(+) = \bar{q}^i_b(-) \otimes \bar{q}^{b(k-1)}_{b(k)}$$

Need to periodically renormalize  $\bar{q}$



- High fidelity

$$\vec{\omega}_{ib}^b \Delta t = \vec{k} \Delta \theta$$

$$\mathfrak{K} = [\vec{k} \times]$$

$$C_b^i(+) = C_b^i(-) [\mathcal{I} + \sin(\Delta\theta)\mathfrak{K} + [1 - \cos(\Delta\theta)] \mathfrak{K}^2] \quad (1)$$

or

$$\bar{q}_b^i(+) = \bar{q}_b^i(-) \otimes \begin{bmatrix} \cos(\frac{\Delta\theta}{2}) \\ \vec{k} \sin(\frac{\Delta\theta}{2}) \end{bmatrix} \quad (2)$$

- Low fidelity

$$C_b^i(+) \approx C_b^i(-) (\mathcal{I} + \Omega_{ib}^b \Delta t) \quad (3)$$

## ② Specific force transformation

- Simply coordinatize the specific force

$$\vec{f}_{ib}^i = C_b^i (+) \vec{f}_{ib}^b \quad (4)$$

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- Assuming that we are in space (i.e., no centrifugal component)

$$\vec{f}_{ib}^i = \vec{a}_{ib}^i - \vec{\gamma}_{ib}^i \quad \vec{a}_{ib}^i = \vec{f}_{ib}^i + \vec{\gamma}_{ib}^i \quad (5)$$

- Thus, by simple numerical integration

$$\vec{v}_{ib}^i (+) = \vec{v}_{ib}^i (-) + \vec{a}_{ib}^i \Delta t \quad (6)$$

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## ④ Position update

- by simple numerical integration

$$\vec{r}_{ib}^i (+) = \vec{r}_{ib}^i (-) + \vec{v}_{ib}^i (-) \Delta t + \vec{a}_{ib}^i \frac{\Delta t^2}{2} \quad (7)$$

$$C_b^i(+) = C_b^i(-) [\mathcal{I} + \sin(\Delta\theta)\hat{\mathbf{R}} + [1 - \cos(\Delta\theta)]\hat{\mathbf{R}}^2]$$

or

$$C_b^i(+) \approx C_b^i(-) (\mathcal{I} + \Omega_{ib}^b \Delta t)$$

or

$$\bar{q}_b^i(+) = \bar{q}_b^i(-) \otimes \begin{bmatrix} \cos(\frac{\Delta\theta}{2}) \\ \vec{k} \sin(\frac{\Delta\theta}{2}) \end{bmatrix}$$

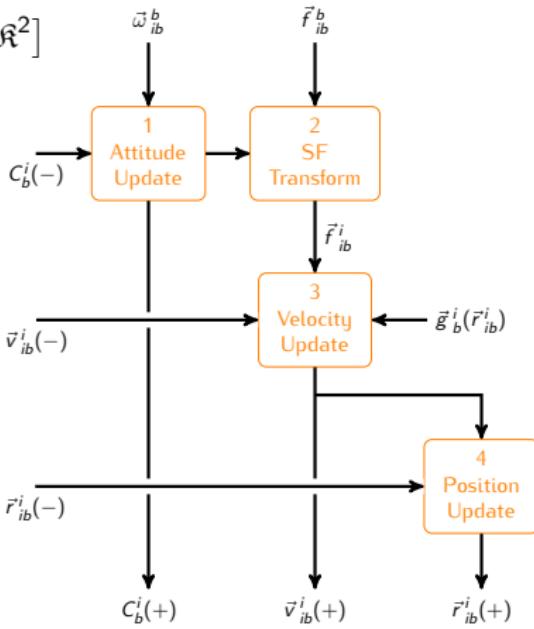
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$$\vec{f}_{ib}^i = C_b^i(+) \vec{f}_{ib}^b$$

$$\vec{a}_{ib}^i = \vec{f}_{ib}^i + \vec{\gamma}_{ib}^i$$

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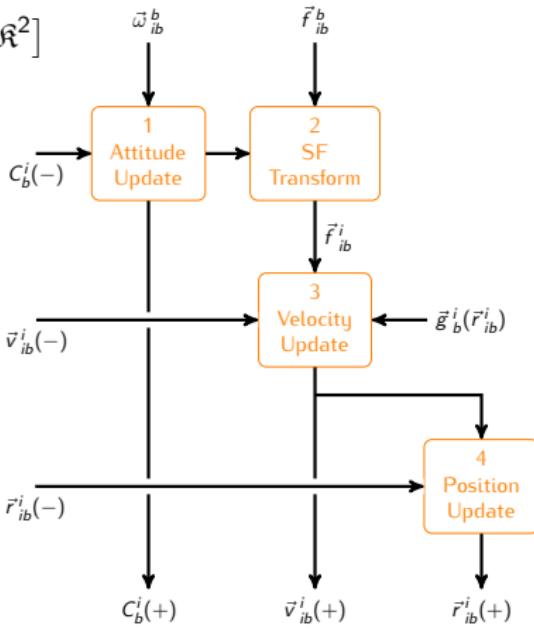
and

$$\vec{f}_{ib}^i = C_b^i(+) \vec{f}_{ib}^b$$

$$\vec{a}_{ib}^i = \vec{f}_{ib}^i + \vec{\gamma}_{ib}^i$$

$$\vec{v}_{ib}^i(+) = \vec{v}_{ib}^i(-) + \vec{a}_{ib}^i \Delta t$$

$$\vec{r}_{ib}^i(+) = \vec{r}_{ib}^i(-) + \vec{v}_{ib}^i(-) \Delta t + \vec{a}_{ib}^i \frac{\Delta t^2}{2}$$



What is the importance of  $\Delta t$ ?

- In continuous time notation

- Attitude:  $\dot{C}_b^i = C_b^i \Omega_{ib}^b$  or  $\dot{\bar{q}}_b^i = \frac{1}{2}[\check{\omega}_{ib}^b \circledast] \bar{q}_b^i(t)$
- Velocity:  $\dot{\vec{v}}_{ib}^i = C_b^i \vec{f}_{ib}^b + \vec{\gamma}_{ib}^i$
- Position:  $\dot{\vec{r}}_{ib}^i = \vec{v}_{ib}^i$

- In State-space notation

$$\begin{bmatrix} \dot{\vec{r}}_{ib}^i \\ \dot{\vec{v}}_{ib}^i \\ \dot{C}_b^i \end{bmatrix} = \begin{bmatrix} \vec{v}_{ib}^i \\ C_b^i \vec{f}_{ib}^b + \vec{\gamma}_{ib}^i \\ C_b^i \Omega_{ib}^b \end{bmatrix} \quad (8)$$

or

$$\begin{bmatrix} \dot{\vec{r}}_{ib}^i \\ \dot{\vec{v}}_{ib}^i \\ \dot{\bar{q}}_b^i \end{bmatrix} = \begin{bmatrix} \vec{v}_{ib}^i \\ C_b^i \vec{f}_{ib}^b + \vec{\gamma}_{ib}^i \\ \frac{1}{2}[\check{\omega}_{ib}^b \circledast] \bar{q}_b^i(t) \end{bmatrix} \quad (9)$$

$$[\bar{q} \otimes] = \begin{bmatrix} q_s & -q_x & -q_y & -q_z \\ q_x & q_s & -q_z & q_y \\ q_y & q_z & q_s & -q_x \\ q_z & -q_y & q_x & q_s \end{bmatrix}$$
$$[\bar{q} \circledast] = \begin{bmatrix} q_s & -q_x & -q_y & -q_z \\ q_x & q_s & q_z & -q_y \\ q_y & -q_z & q_s & q_x \\ q_z & q_y & -q_x & q_s \end{bmatrix}$$