EE 570: Location and Navigation Error Mechanization (Tangential)

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Tangential Mechanization — Continuous Case



In continuous time notation

• Attitude:
$$\dot{C}_b^t = C_b^t \Omega_{tb}^b$$

• Velocity:
$$\dot{\vec{v}}_{tb}^{t} = C_b^t \vec{f}_{ib}^{b} + \vec{g}_b^t - 2\Omega_{ie}^t \vec{v}_{tb}^t$$

• Position:
$$\dot{\vec{r}}_{ib}^t = \vec{v}_{tb}^t$$

In State-space notation

$$\vec{\omega}_{ib}^b = \vec{\omega}_{ie}^b + \vec{\omega}_{tb}^b$$

$$\Omega^b_{tb} = \Omega^b_{ib} - \Omega^b_{ie}$$

$$\begin{bmatrix} \dot{\vec{r}}_{tb}^t \\ \dot{\vec{v}}_{tb}^t \\ \dot{C}_b^t \end{bmatrix} = \begin{bmatrix} \vec{v}_{tb}^t \\ C_b^t \vec{f}_{ib}^b + \vec{g}_b^t - 2\Omega_{ie}^t \vec{v}_{tb}^t \\ C_b^t \Omega_{tb}^b \end{bmatrix}$$
(1)

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$$\vec{r}_{ib}^t = \vec{v}_{tb}^t$$

In State-space notation

$$\vec{\omega}_{ib}^b = \vec{\omega}_{ie}^b + \vec{\omega}_{tb}^b$$

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$$\begin{bmatrix} \dot{\vec{r}}_{tb}^t \\ \dot{\vec{v}}_{tb}^t \\ \dot{C}_b^t \end{bmatrix} = \begin{bmatrix} \vec{v}_{tb}^t \\ C_b^t \vec{f}_{ib}^b + \vec{g}_b^t - 2\Omega_{ie}^t \vec{v}_{tb}^t \\ C_b^t \Omega_{tb}^b \end{bmatrix}$$
(1)

Question

What is the effect of sensor noise in the measurements $\vec{\omega}_{ib}^b$ and \vec{f}_{ib}^b on the navigation solution position, velocity and attitude?



$$\dot{C}_b^t = C_b^t \Omega_{tb}^b = C_b^t (\Omega_{ib}^b - \Omega_{ie}^b) = \frac{d}{dt} \left[(\mathcal{I} + [\delta \vec{\psi}_{tb}^t \times]) \hat{C}_b^t \right] = 0$$



$$\dot{C}^t_b = C^t_b \Omega^b_{tb} = C^t_b (\Omega^b_{ib} - \Omega^b_{ie}) = \frac{d}{dt} \left[(\mathcal{I} + [\delta \vec{\psi}^t_{tb} \times]) \hat{C}^t_b \right] =$$

$$(\mathcal{I} + [\delta \vec{\psi}_{tb}^t \times]) \hat{C}_b^t \Omega_{tb}^b = [\delta \dot{\vec{\psi}}_{tb}^t \times] \hat{C}_b^t + (\mathcal{I} + [\delta \vec{\psi}_{tb}^t \times]) \dot{C}_b^t =$$



$$\dot{C}^t_b = C^t_b \Omega^b_{tb} = C^t_b (\Omega^b_{ib} - \Omega^b_{ie}) = \frac{d}{dt} \left[(\mathcal{I} + [\delta \vec{\psi}^t_{tb} \times]) \hat{C}^t_b \right] =$$

$$(\mathcal{I} + [\delta \vec{\psi}_{tb}^t \times]) \hat{C}_b^t \Omega_{tb}^b = [\delta \vec{\psi}_{tb}^t \times] \hat{C}_b^t + (\mathcal{I} + [\delta \vec{\psi}_{tb}^t \times]) \hat{C}_b^t = (\mathcal{I} + [\delta \vec{\psi}_{tb}^t \times]) \hat{C}_b^t (\hat{\Omega}_{tb}^b + \delta \Omega_{ib}^b - \delta \Omega_{ie}^b) =$$

Attitude Velocity Gravity Position Summary



$$\dot{C}^t_b = C^t_b \Omega^b_{tb} = C^t_b (\Omega^b_{ib} - \Omega^b_{ie}) = \frac{d}{dt} \left[(\mathcal{I} + [\delta \vec{\psi}^t_{tb} \times]) \hat{C}^t_b \right] =$$

$$\begin{split} (\mathcal{I} + [\delta\vec{\psi}_{tb}^t \times])\hat{C}_b^t \Omega_{tb}^b &= [\delta\dot{\vec{\psi}}_{tb}^t \times]\hat{C}_b^t + (\mathcal{I} + [\delta\vec{\psi}_{tb}^t \times])\dot{\hat{C}}_b^t = \\ (\mathcal{I} + [\delta\vec{\psi}_{tb}^t \times])\hat{C}_b^t (\hat{\Omega}_{tb}^b + \delta\Omega_{ib}^b - \delta\Omega_{ie}^b) &= \\ (\mathcal{I} + [\delta\vec{\psi}_{tb}^t \times])\hat{C}_b^t \hat{\Omega}_{tb}^b + \hat{C}_b^t (\delta\Omega_{ib}^b - \delta\Omega_{ie}^b) &= \\ &: (\mathcal{I} + [\delta\vec{\psi}_{tb}^t \times])\hat{C}_b^t \hat{\Omega}_{tb}^b + \hat{C}_b^t (\delta\Omega_{ib}^b - \delta\Omega_{ie}^b) &= \\ &: (\delta\vec{\psi}_{tb}^t \times]\delta\Omega_{tb}^b \approx 0 \end{split}$$



$$\begin{split} \dot{C}_{b}^{t} &= C_{b}^{t}\Omega_{tb}^{b} = C_{b}^{t}(\Omega_{ib}^{b} - \Omega_{ie}^{b}) = \frac{d}{dt}\left[(\mathcal{I} + [\delta\vec{\psi}_{tb}^{t}\times])\hat{C}_{b}^{t} \right] = \\ & (\mathcal{I} + [\delta\vec{\psi}_{tb}^{t}\times])\hat{C}_{b}^{t}\Omega_{tb}^{b} = [\delta\dot{\vec{\psi}}_{tb}^{t}\times]\hat{C}_{b}^{t} + (\mathcal{I} + [\delta\vec{\psi}_{tb}^{t}\times])\hat{C}_{b}^{t}) = \\ & (\mathcal{I} + [\delta\vec{\psi}_{tb}^{t}\times])\hat{C}_{b}^{t}(\hat{\Omega}_{tb}^{b} + \delta\Omega_{ib}^{b} - \delta\Omega_{ie}^{b}) = \\ & (\mathcal{I} + [\delta\vec{\psi}_{tb}^{t}\times])\hat{C}_{b}^{t}(\hat{\Omega}_{tb}^{b} + \hat{C}_{b}^{t}(\delta\Omega_{ib}^{b} - \delta\Omega_{ie}^{b}) = \end{split}$$



$$\dot{C}^t_b = C^t_b \Omega^b_{tb} \ = C^t_b (\Omega^b_{ib} - \Omega^b_{ie}) = \frac{d}{dt} \left[(\mathcal{I} + [\delta \vec{\psi}^t_{tb} \times]) \hat{C}^t_b \right] =$$

$$\begin{split} (\mathcal{I} + [\delta\vec{\psi}_{tb}^t\times])\hat{C}_b^t\Omega_{tb}^b &= [\delta\vec{\psi}_{tb}^t\times]\hat{C}_b^t + (\mathcal{I} + [\delta\vec{\psi}_{tb}^t\times])\hat{C}_b^t = \\ (\mathcal{I} + [\delta\vec{\psi}_{tb}^t\times])\hat{C}_b^t(\hat{\Omega}_{tb}^b + \delta\Omega_{ib}^b - \delta\Omega_{ie}^b) &= \\ (\mathcal{I} + [\delta\vec{\psi}_{tb}^t\times])\hat{C}_b^t\hat{\Omega}_{tb}^b + \hat{C}_b^t(\delta\Omega_{ib}^b - \delta\Omega_{ie}^b) &= \end{split}$$

$$[\delta \dot{\vec{\psi}}_{tb}^t \times] = \hat{C}_b^t (\delta \Omega_{ib}^b - \delta \Omega_{ie}^b) \hat{C}_t^b = [\hat{C}_b^t (\delta \vec{\omega}_{ib}^b - \delta \vec{\omega}_{ie}^b) \times]$$
(2)

Attitude Velocity Gravity Position Summary



$$\dot{C}^t_b = C^t_b \Omega^b_{tb} \ = C^t_b (\Omega^b_{ib} - \Omega^b_{ie}) = \frac{d}{dt} \left[(\mathcal{I} + [\delta \vec{\psi}^t_{tb} \times]) \hat{C}^t_b \right] =$$

$$\begin{split} (\mathcal{I} + [\delta\vec{\psi}_{tb}^t\times])\hat{C}_b^t\Omega_{tb}^b &= [\delta\vec{\psi}_{tb}^t\times]\hat{C}_b^t + (\mathcal{I} + [\delta\vec{\psi}_{tb}^t\times])\hat{C}_b^t = \\ (\mathcal{I} + [\delta\vec{\psi}_{tb}^t\times])\hat{C}_b^t(\hat{\Omega}_{tb}^b + \delta\Omega_{ib}^b - \delta\Omega_{ie}^b) &= \\ (\mathcal{I} + [\delta\vec{\psi}_{tb}^t\times])\hat{C}_b^t\hat{\Omega}_{tb}^b + \hat{C}_b^t(\delta\Omega_{ib}^b - \delta\Omega_{ie}^b) &= \end{split}$$

$$[\delta \vec{\psi}_{tb}^{t} \times] = \hat{C}_{b}^{t} (\delta \Omega_{ib}^{b} - \delta \Omega_{ie}^{b}) \hat{C}_{t}^{b} = [\hat{C}_{b}^{t} (\delta \vec{\omega}_{ib}^{b} - \delta \vec{\omega}_{ie}^{b}) \times]$$
(2)

$$\delta \dot{\psi}_{tb}^{t} = \hat{C}_{b}^{t} (\delta \vec{\omega}_{ib}^{b} - \delta \vec{\omega}_{ie}^{b}) \tag{3}$$

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Tangential Attitude Error (cont.)



$$\begin{split} \delta \dot{\vec{\psi}}_{tb}^t &= \hat{C}_b^t (\delta \vec{\omega}_{ib}^b - \delta \vec{\omega}_{ie}^b) \\ &= \hat{C}_b^t \delta \vec{\omega}_{ib}^b - \hat{C}_b^t (\vec{\omega}_{ie}^b - \hat{\vec{\omega}}_{ie}^b) \\ &= \hat{C}_b^t \delta \vec{\omega}_{ib}^b - (\hat{C}_b^t C_t^b - \mathcal{I}) \vec{\omega}_{ie}^t \\ &= \hat{C}_b^t \delta \vec{\omega}_{ib}^b + \delta \vec{\psi}_{tb}^t \times \vec{\omega}_{ie}^t \end{split}$$

Attitude Velocity Gravity Position Summary

Tangential Attitude Error (cont.)



$$\begin{split} \delta \dot{\vec{\psi}}_{tb}^t &= \hat{C}_b^t (\delta \vec{\omega}_{ib}^b - \delta \vec{\omega}_{ie}^b) \\ &= \hat{C}_b^t \delta \vec{\omega}_{ib}^b - \hat{C}_b^t (\vec{\omega}_{ie}^b - \hat{\vec{\omega}}_{ie}^b) \\ &= \hat{C}_b^t \delta \vec{\omega}_{ib}^b - (\hat{C}_b^t C_t^b - \mathcal{I}) \vec{\omega}_{ie}^t \\ &= \hat{C}_b^t \delta \vec{\omega}_{ib}^b + \delta \vec{\psi}_{tb}^t \times \vec{\omega}_{ie}^t \end{split}$$

$$\delta \dot{\vec{\psi}}_{tb}^t = \hat{C}_b^t \delta \vec{\omega}_{ib}^b - \vec{\Omega}_{ie}^t \delta \vec{\psi}_{tb}^t \tag{4}$$

Attitude Velocity Gravity Position Summary

Velocity



$$\dot{\vec{v}}_{tb}^{t} = C_b^t \vec{f}_{ib}^{b} + \vec{g}_b^t - 2\Omega_{ie}^t \vec{v}_{tb}^t$$
 (5)

$$\dot{\vec{v}}_{tb}^{t} = \hat{C}_{b}^{t} \dot{\vec{f}}_{ib}^{b} + \hat{\vec{g}}_{b}^{t} - 2\Omega_{ie}^{t} \dot{\vec{v}}_{tb}^{t}
= (\mathcal{I} - [\delta \vec{\psi}_{tb}^{t} \times]) C_{b}^{t} (\vec{f}_{ib}^{b} - \delta \vec{f}_{ib}^{b}) + \hat{\vec{g}}_{b}^{t} - 2\Omega_{ie}^{t} \hat{\vec{v}}_{tb}^{t}$$
(6)

Velocity



$$\dot{\vec{v}}_{tb}^{t} = C_{b}^{t} \vec{f}_{ib}^{b} + \vec{g}_{b}^{t} - 2\Omega_{ie}^{t} \vec{v}_{tb}^{t}$$
 (5)

$$\dot{\hat{\vec{v}}}_{tb}^{t} = \hat{C}_{b}^{t} \hat{\vec{f}}_{ib}^{b} + \hat{\vec{g}}_{b}^{t} - 2\Omega_{ie}^{t} \hat{\vec{v}}_{tb}^{t}
= (\mathcal{I} - [\delta \vec{\psi}_{tb}^{t} \times]) C_{b}^{t} (\vec{f}_{ib}^{b} - \delta \vec{f}_{ib}^{b}) + \hat{\vec{g}}_{b}^{t} - 2\Omega_{ie}^{t} \hat{\vec{v}}_{tb}^{t}$$
(6)

$$\begin{split} \delta \vec{v}_{tb}^t &= \dot{\vec{v}}_{tb}^t - \dot{\vec{v}}_{tb}^t = [\delta \vec{\psi}_{tb}^t \times] C_b^t \vec{f}_{ib}^b + \hat{C}_b^t \delta \vec{f}_{ib}^b + \delta \vec{g}_b^t - 2 \Omega_{ie}^t \delta \vec{v}_{tb}^t \\ &= [\delta \vec{\psi}_{tb}^t \times] \hat{C}_b^t \hat{\vec{f}}_{ib}^b + \hat{C}_b^t \delta \vec{f}_{ib}^b + \delta \vec{g}_b^t - 2 \Omega_{ie}^t \delta \vec{v}_{tb}^t \end{split}$$

Velocity



$$\dot{\vec{v}}_{tb}^{t} = C_b^t \vec{f}_{ib}^{b} + \vec{g}_b^t - 2\Omega_{ie}^t \vec{v}_{tb}^t$$
 (5)

$$\frac{\dot{\hat{v}}_{tb}^{t} = \hat{C}_{b}^{t} \hat{\vec{f}}_{ib}^{b} + \hat{\vec{g}}_{b}^{t} - 2\Omega_{ie}^{t} \hat{\vec{v}}_{tb}^{t}}{= (\mathcal{I} - [\delta \vec{\psi}_{tb}^{t} \times]) C_{b}^{t} (\vec{f}_{ib}^{b} - \delta \vec{f}_{ib}^{b}) + \hat{\vec{g}}_{b}^{t} - 2\Omega_{ie}^{t} \hat{\vec{v}}_{tb}^{t}} \tag{6}$$

$$\begin{split} \delta \vec{v}_{tb}^t &= \dot{\vec{v}}_{tb}^t - \dot{\vec{v}}_{tb}^t = [\delta \vec{\psi}_{tb}^t \times] C_b^t \vec{f}_{ib}^b + \hat{C}_b^t \delta \vec{f}_{ib}^b + \delta \vec{g}_b^t - 2\Omega_{ie}^t \delta \vec{v}_{tb}^t \\ &= [\delta \vec{\psi}_{tb}^t \times] \hat{C}_b^t \hat{\vec{f}}_{ib}^b + \hat{C}_b^t \delta \vec{f}_{ib}^b + \delta \vec{g}_b^t - 2\Omega_{ie}^t \delta \vec{v}_{tb}^t \end{split}$$

$$\delta \vec{\mathbf{v}}_{tb}^{t} = -[\hat{C}_{b}^{t} \hat{\vec{f}}_{ib}^{b} \times]\delta \vec{\psi}_{tb}^{t} + \hat{C}_{b}^{t} \delta \vec{f}_{ib}^{b} + \delta \vec{\mathbf{g}}_{b}^{t} - 2\Omega_{ie}^{t} \hat{\vec{\mathbf{v}}}_{tb}^{t}$$
 (7)



Gravity Error



$$\delta \vec{g}_b^t \approx \hat{C}_e^t \frac{2g_0(\hat{L}_b)}{r_{eS}^e(\hat{L}_b)} \frac{\hat{\vec{r}}_{eb}^e}{|\hat{\vec{r}}_{eb}^e|^2} (\hat{\vec{r}}_{eb}^e)^T \hat{C}_t^e \delta \vec{r}_{tb}^t$$
(8)

Position



$$\dot{\vec{r}}_{tb}^{t} = \vec{v}_{tb}^{t} \tag{9}$$



$$\dot{\vec{r}}_{tb}^{t} = \vec{v}_{tb}^{t} \tag{9}$$

$$\delta \dot{\vec{r}}_{tb}^{t} = \delta \vec{v}_{tb}^{t} \tag{10}$$



$$\begin{pmatrix} \delta \dot{\vec{\psi}}_{tb}^{t} \\ \delta \dot{\vec{v}}_{tb}^{t} \\ \delta \dot{\vec{r}}_{tb}^{t} \end{pmatrix} = \begin{bmatrix} -\Omega_{ie}^{t} & 0_{3\times3} & 0_{3\times3} \\ -[\hat{C}_{b}^{t}\hat{\vec{f}}_{ib}^{b}\times] & -2\Omega_{ie}^{t} & \hat{C}_{e}^{t} \frac{2g_{0}(\hat{L}_{b})}{r_{es}^{e}(\hat{L}_{b})} \frac{\hat{r}_{eb}^{e}}{|\hat{r}_{eb}^{e}|^{2}} (\hat{\vec{r}}_{eb}^{e})^{T} \hat{C}_{t}^{e} \\ 0_{3\times3} & \mathcal{I}_{3\times3} & 0_{3\times3} \end{bmatrix} \begin{pmatrix} \delta \vec{\psi}_{tb}^{t} \\ \delta \vec{r}_{tb}^{t} \end{pmatrix} +$$

$$\begin{bmatrix} 0 & \hat{C}_{b}^{t} \\ \hat{C}_{b}^{t} & 0 \\ 0 & 0 \end{bmatrix} \begin{pmatrix} \delta \vec{f}_{ib}^{b} \\ \delta \vec{\omega}_{ib}^{b} \end{pmatrix}$$

(11)

Attitude Velocity Gravity Position **Summary**



Truth value

$$\vec{x}$$

Measured value

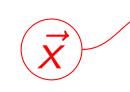
• Estimated or computed value

$$\delta \vec{x} = \vec{x} - \hat{\vec{x}}$$



Nothing above

Truth value



Measured value

• Estimated or computed value

$$\delta \vec{x} = \vec{x} - \hat{\vec{x}}$$



Truth value

Measured value



• Estimated or computed value

$$\hat{\vec{x}}$$

$$\delta \vec{x} = \vec{x} - \hat{\vec{x}}$$



Truth value

 \vec{x}

Measured value

 $\tilde{\bar{x}}$

• Estimated or computed value



Error

$$\delta \vec{x} = \vec{x} - \hat{\vec{x}}$$



Truth value

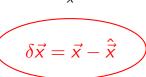
$$\vec{x}$$

Measured value

$$\bar{\bar{x}}$$

• Estimated or computed value







Given a non-linear system $\dot{\vec{x}} = f(\vec{x}, t)$





Given a non-linear system $\vec{x} = f(\vec{x}, t)$ Let's assume we have an estimate of \vec{x} , i.e., $\hat{\vec{x}}$ such that $\vec{x} = \hat{\vec{x}} + \delta \vec{x}$

$$\dot{\vec{x}} = \dot{\vec{x}} + \delta \dot{\vec{x}} = f(\hat{\vec{x}} + \delta \vec{x}, t) \tag{12}$$



Given a non-linear system $\dot{\vec{x}} = f(\vec{x}, t)$ Let's assume we have an estimate of \vec{x} , i.e., $\dot{\vec{x}}$ such that $\vec{x} = \dot{\vec{x}} + \delta \vec{x}$

$$\dot{\vec{x}} = \dot{\vec{x}} + \delta \dot{\vec{x}} = f(\hat{\vec{x}} + \delta \vec{x}, t)$$
 (12)

Using Taylor series expansion

$$f(\hat{\vec{x}} + \delta \vec{x}, t) = \dot{\hat{\vec{x}}} + \delta \dot{\vec{x}} = f(\hat{\vec{x}}, t) + \frac{\partial f(\vec{x}, t)}{\partial \vec{x}} \Big|_{\vec{x} = \hat{\vec{x}}} \delta \vec{x} + H.O.T$$

$$\approx \dot{\hat{\vec{x}}} + \frac{\partial f(\vec{x}, t)}{\partial \vec{x}} \Big|_{\vec{x} = \hat{\vec{x}}} \delta \vec{x}$$



Given a non-linear system $\dot{\vec{x}} = f(\vec{x}, t)$ Let's assume we have an estimate of \vec{x} , i.e., $\hat{\vec{x}}$ such that $\vec{x} = \hat{\vec{x}} + \delta \vec{x}$

$$\dot{\vec{x}} = \dot{\vec{x}} + \delta \dot{\vec{x}} = f(\hat{\vec{x}} + \delta \vec{x}, t)$$
 (12)

Using Taylor series expansion

$$f(\hat{\vec{x}} + \delta \vec{x}, t) = \dot{\hat{\vec{x}}} + \delta \dot{\vec{x}} = \underbrace{f(\hat{\vec{x}}, t)}_{=\hat{\vec{x}}} + \frac{\partial f(\vec{x}, t)}{\partial \vec{x}} \Big|_{\vec{x} = \hat{\vec{x}}} \delta \vec{x} + H.O.T$$

$$\approx \dot{\hat{\vec{x}}} + \frac{\partial f(\vec{x}, t)}{\partial \vec{x}} \Big|_{\vec{x} = \hat{\vec{x}}} \delta \vec{x}$$



Given a non-linear system $\dot{\vec{x}} = f(\vec{x}, t)$ Let's assume we have an estimate of \vec{x} , i.e., $\dot{\vec{x}}$ such that $\vec{x} = \dot{\vec{x}} + \delta \vec{x}$

$$\dot{\vec{x}} = \dot{\vec{x}} + \delta \dot{\vec{x}} = f(\hat{\vec{x}} + \delta \vec{x}, t)$$
 (12)

Using Taylor series expansion

$$f(\hat{\vec{x}} + \delta \vec{x}, t) = \dot{\hat{\vec{x}}} + \delta \dot{\vec{x}} = f(\hat{\vec{x}}, t) + \frac{\partial f(\vec{x}, t)}{\partial \vec{x}} \Big|_{\vec{x} = \hat{\vec{x}}} \delta \vec{x} + H.O.T$$

$$\approx \dot{\hat{\vec{x}}} + \frac{\partial f(\vec{x}, t)}{\partial \vec{x}} \Big|_{\vec{x} = \hat{\vec{x}}} \delta \vec{x}$$

$$\Rightarrow \delta \dot{\vec{x}} \approx \left. \frac{\partial f(\vec{x}, t)}{\partial \vec{x}} \right|_{\vec{x} = \hat{\vec{x}}} \delta \vec{x} \tag{13}$$

Actual Measurements



Initially the accelerometer and gyroscope measurements, $\tilde{\vec{f}}_{ib}^{\ b}$ and $\tilde{\vec{\omega}}_{ib}^{\ b}$, respectively, will be modeled as

$$\tilde{\vec{f}}_{ib}^b = \vec{f}_{ib}^b + \Delta \vec{f}_{ib}^b \tag{14}$$

$$\tilde{\vec{\omega}}_{ib}^{b} = \vec{\omega}_{ib}^{b} + \Delta \vec{\omega}_{ib}^{b} \tag{15}$$

where $\vec{f}_{ib}^{\ b}$ and $\vec{\omega}_{ib}^{\ b}$ are the specific force and angular rates, respectively; and $\Delta \vec{f}_{ib}^{\ b}$ and $\Delta \vec{\omega}_{ib}^{\ b}$ represents the errors. In later lectures we will discuss more detailed description of these errors.

Actual Measurements



Initially the accelerometer and gyroscope measurements, \tilde{f}^b_{ib} and $\tilde{\vec{\omega}}^b_{ib}$ respectively, will be modeled as

$$\tilde{\vec{f}}_{ib}^{b} = \vec{f}_{ib}^{b} + \Delta \vec{f}_{ib}^{b}$$
these terms may
$$\tilde{\vec{\omega}}_{ib}^{b} = \vec{\omega}_{ib}^{b} + \Delta \vec{\omega}_{ib}^{b}$$
be expanded further
$$(14)$$

$$\tilde{\vec{\omega}}_{ib}^{b} = \vec{\omega}_{ib}^{b} + \left(\Delta \vec{\omega}_{ib}^{b}\right)^{\text{be expanded further}} \tag{15}$$

where $\vec{f}_{ib}^{\ b}$ and $\vec{\omega}_{ib}^{\ b}$ are the specific force and angular rates, respectively; and $\Delta \vec{f}_{ib}^{\ b}$ and $\Delta \vec{\omega}_{ib}^{\ b}$ represents the errors. In later lectures we will discuss more detailed description of these errors.

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Accelerometers

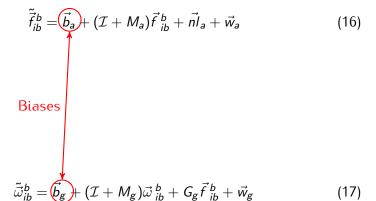
$$\tilde{\vec{f}}_{ib}^{b} = \vec{b}_{a} + (\mathcal{I} + M_{a})\vec{f}_{ib}^{b} + n\vec{l}_{a} + \vec{w}_{a}$$
 (16)

Gyroscopes

$$\tilde{\vec{\omega}}_{ib}^{b} = \vec{b}_{g} + (\mathcal{I} + M_{g})\vec{\omega}_{ib}^{b} + G_{g}\vec{f}_{ib}^{b} + \vec{w}_{g}$$
 (17)



Accelerometers

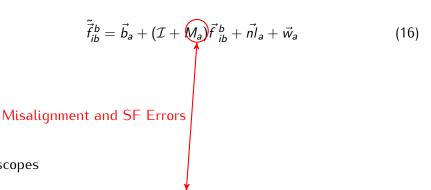


Gyroscopes

asic Definitions Linearization Inertial Measurements Attitude Error Estimates of Sensor Measurements



Accelerometers



Gyroscopes

$$\tilde{\vec{\omega}}_{ib}^{b} = \vec{b}_{g} + (\mathcal{I} + (M_{g}))\vec{\omega}_{ib}^{b} + G_{g}\vec{f}_{ib}^{b} + \vec{w}_{g}$$
 (17)



Accelerometers

$$\tilde{\vec{f}}_{ib}^{b} = \vec{b}_{a} + (\mathcal{I} + M_{a})\vec{f}_{ib}^{b} + (\vec{nl}_{a})\vec{r}_{ib} + \vec{w}_{a}$$
(16)

Non-linearity

Gyroscopes

$$\tilde{\vec{\omega}}_{ib}^{b} = \vec{b}_{g} + (\mathcal{I} + M_{g})\vec{\omega}_{ib}^{b} + G_{g}\vec{f}_{ib}^{b} + \vec{w}_{g}$$
 (17)



Accelerometers

$$\tilde{\vec{f}}_{ib}^{b} = \vec{b}_{a} + (\mathcal{I} + M_{a})\vec{f}_{ib}^{b} + \vec{nl}_{a} + \vec{w}_{a}$$
 (16)

Gyroscopes

$$\tilde{\vec{\omega}}_{ib}^{b} = \vec{b}_{g} + (\mathcal{I} + M_{g})\vec{\omega}_{ib}^{b} + G_{g}\vec{f}_{ib}^{b} + \vec{w}_{g}$$
 (17)

Basic Definitions Linearization Inertial Measurements Attitude Error Estimates of Sensor Measurements



Accelerometers

$$\tilde{\vec{f}}_{ib}^{b} = \vec{b}_{a} + (\mathcal{I} + M_{a})\vec{f}_{ib}^{b} + \vec{nl}_{a} + \vec{w}_{a}$$
(16)

Gyroscopes

$$\tilde{\vec{\omega}}_{ib}^{b} = \vec{b}_{g} + (\mathcal{I} + M_{g})\vec{\omega}_{ib}^{b} + G_{g}\vec{f}_{ib}^{b} + \vec{w}_{g}$$
 (17)

Pos, Vel, Force and Angular Rate Errors



Position error

$$\delta \vec{r}_{\beta b}^{\gamma} = \vec{r}_{\beta b}^{\gamma} - \hat{\vec{r}}_{\beta b}^{\gamma} \tag{18}$$

Velocity error

$$\delta \vec{\mathbf{v}}_{\beta b}^{\gamma} = \vec{\mathbf{v}}_{\beta b}^{\gamma} - \hat{\vec{\mathbf{v}}}_{\beta b}^{\gamma} \tag{19}$$

Specific force errors

$$\delta \vec{f}_{ib}^{\ b} = \vec{f}_{ib}^{\ b} - \hat{\vec{f}}_{ib}^{\ b} \tag{20}$$

$$\Delta_{e}\vec{f}_{ib}^{\ b} = \Delta\vec{f}_{ib}^{\ b} - \Delta\hat{\vec{f}}_{ib}^{\ b} = -\delta\vec{f}_{ib}^{\ b} \tag{21}$$

Angular rate errors

$$\delta\vec{\omega}_{ib}^{\ b} = \vec{\omega}_{ib}^{\ b} - \hat{\vec{\omega}}_{ib}^{\ b} \tag{22}$$

$$\Delta_e \vec{\omega}_{ib}^{\ b} = \Delta \vec{\omega}_{ib}^{\ b} - \Delta \hat{\vec{\omega}}_{ib}^{\ b} = -\delta \vec{\omega}_{ib}^{\ b} \tag{23}$$

Attitude Error Definition



Define

$$\delta C_b^{\gamma} = C_b^{\gamma} \hat{C}_{\gamma}^b = e^{[\delta \vec{\psi}_{\gamma b}^{\gamma} \times]} \approx \mathcal{I} + [\delta \vec{\psi}_{\gamma b}^{\gamma} \times]$$
 (24)

This is the error in attitude resulting from errors in estimating the angular rates.

Attitude Error Properties



The attitude error is a multiplicative small angle transformation from the actual frame to the computed frame

$$\hat{C}_{b}^{\gamma} = (\mathcal{I} - [\delta \vec{\psi}_{\gamma b}^{\gamma} \times]) C_{b}^{\gamma} \tag{25}$$

Similarly,

$$C_b^{\gamma} = (\mathcal{I} + [\delta \vec{\psi}_{\gamma b}^{\gamma} \times]) \hat{C}_b^{\gamma}$$
 (26)

Specific Force and Agnular Rates



Similarly we can attempt to estimate the specific force and angular rate by applying correction based on our estimate of the error.

$$\hat{\vec{f}}_{ib}^{\ b} = \tilde{\vec{f}}_{ib}^{\ b} - \Delta \hat{\vec{f}}_{ib}^{\ b} \tag{27}$$

$$\hat{\vec{\omega}}_{ib}^{b} = \tilde{\vec{\omega}}_{ib}^{b} - \Delta \hat{\vec{\omega}}_{ib}^{b} \tag{28}$$

where $\hat{\vec{f}}_{ib}^{\ b}$ and $\hat{\vec{\omega}}_{ib}^{\ b}$ are the accelerometer and gyroscope estimated calibration values, respectively.